Armed Services Technical Information Agency

Because of our limited supply, you are requested to return this copy WHEN IT HAS SERVED YOUR PURPOSE so that it may be made available to other requesters. Your cooperation will be appreciated.

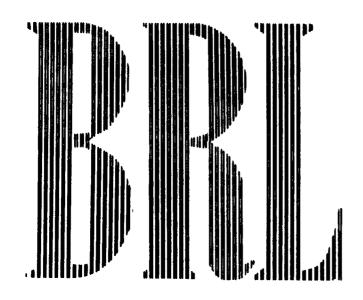
AD

NOTICE: WHEN GOVERNMENT OR OTHER DRAWINGS, SPECIFICATIONS OR OTHER DATA ARE USED FOR ANY PURPOSE OTHER THAN IN CONNECTION WITH A DEFINITELY RELATED GOVERNMENT PROCUREMENT OPERATION, THE U. S. GOVERNMENT THEREBY INCURS NO RESPONSIBILITY, NOR ANY OBLIGATION WHATSOEVER; AND THE FACT THAT THE GOVERNMENT MAY HAVE FORMULATED, FURNISHED, OR IN ANY WAY SUPPLIED THE SAID DRAWINGS, SPECIFICATIONS, OR OTHER DATA IS NOT TO BE REGARDED BY IMPLICATION OR OTHERWISE AS IN ANY MANNER LICENSING THE HOLDER OR ANY OTHER PERSON OR CORPORATION, OR CONVEYING ANY RIGHTS OR PERMISSION TO MANUFACTURE, USE OR SELL ANY PATENTED INVENTION THAT MAY IN ANY WAY BE RELATED THERETO.

Reproduced by DOCUMENT SERVICE CENTER KNOTT BUILDING, DAYTON, 2, 0 HIO

UNCLASSIFIED





REPORT No. 888

The Deflection of A Continuous Beam Produced By The Vertical Motion of A Support Point

TURNER L. SMITH

DEPARTMENT OF THE ARMY PROJECT No. 503-06-004
ORDNANCE RESEARCH AND DEVELOPMENT PROJECT No. TB3-0118H

BALLISTIC RESEARCH LABORATORIES



ABERDEEN PROVING GROUND, MARYLAND

BALLISTIC RESEARCH LABORATORIES

REPORT NO. 888

December 1953

THE DEFLECTION OF A CONTINUOUS BEAM PRODUCED BY THE VERTICAL MOTION OF A SUPPORT POINT

Turner L. Smith

Department of the Army Project No. 503-06-004 Ordnance Research and Development Project No. TB3-0118H

ABERDEEN PROVING GROUND, MARYLAND

BALLISTIC RESEARCH LABORATORIES

REPORT NO. 888

TISmith/ekb Aberdeen Proving Ground, Md. December 1953

THE DEFLECTION OF A CONTINUOUS BEAM PRODUCED BY THE VERTICAL MOTION OF A SUPPORT POINT

ABSTRACT

The elastic curve for a uniform continuous beam, simply restrained to have zero deflection at four or more equidistant points, satisfies a second order linear finite difference equation. The general solution for the elastic curve is found to be the sum of a decreasing wave and an increasing wave. By a combination of these, the elastic curve produced by moving any one support point is obtained, including the cases where the support point which is moved is at one end or near one end. These results are useful in obtaining a logical procedure for correcting supersonic nozzles.

support points, the right member of equation (1.1) is a linear function of x. Hence,

Theorem 1. The deflection y is continuous with continuous slope and curvature (y' and y''), and each segment between support points is a cubic.

DERIVATION OF THE DIFFERENCE EQUATION

Let the deflection of the beam be zero at four successive support points, P_1 , P_2 , P_3 , P_4 , and take individual axis systems (x_i, y_i) for the three included segments of the elastic curve.

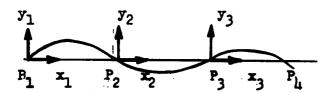


Figure 1.

Then, since each segment is a cubic which vanishes at its origin, the equations of these segments are

$$Y_i = A_i x_i^3 + B_i x_i^2 + C_i x_i, i = 1, 2, 3$$

Continuity at P_2 , P_3 , and P_4 gives three equations found by letting $x_1 = 1$.

(2.1)
$$A_1 + B_1 + C_1 = 0$$

$$A_2 + B_2 + C_2 = 0$$

$$A_3 + B_3 + C_3 = 0$$

Continuity of y^1 and y^m at P_2 and P_3 give

$$3A_1 + 2B_1 + C_1 = C_2$$

$$3A_2 + 2B_2 + C_2 = C_3$$

$$6A_1 + 2B_1 = 2B_2$$

$$6A_2 + 2B_2 = 2B_3$$

These seven equations for nine unknowns leave two degrees of freedom. The elastic curve would be completely determined, for example, if the slopes were prescribed at both ends P_1 and P_{ll} . From seven equations we can eliminate six unknowns and obtain a single equation in the remaining three. In this manner, it is found as a consequence of the seven equations (2.1) and (2.2) that

(2.3)
$$A_{1} + \mu A_{2} + A_{3} = 0$$

$$B_{1} + \mu B_{2} + B_{3} = 0$$

$$C_{1} + \mu C_{2} + C_{3} = 0$$

Since the moment is

$$M_i = EIy_i^* = EI(6A_ix_i + 2B_i)$$

and hence the moment at x = 0 is M = 2B EI, it follows that

$$M_1 + \mu M_2 + M_3 = 0$$

which is the well-known "three-moment equation". Similarly, by letting $x_4 = 1$, we get

$$M_2 + \mu_3 + M_{\mu} = 0$$

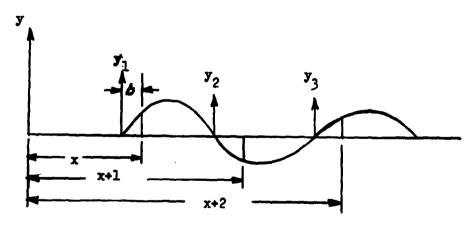


Figure 2

If we form the function

$$(2.4) y(x+2) + Ly(x+1) + y(x)$$

involving heights at corresponding points in three successive segments, this may be written

$$y_3(b) + \mu y_2(b) + y_1(b) = (A_3 + \mu A_2 + A_1)b^3 + (B_3 + \mu B_2 + B_1)b^2 + (C_3 + \mu C_2 + C_1)b$$

which vanishes on account of equations (2.3) for all values of b in the range $0 \le b \le 1$. Hence, the expression (2.4) is zero for any x in the interval $P_1 - P_2$.

Since y(x) is of class C^n , the first and second derivative of the expression (2.4) also vanishes. Moreover, if the elastic curve has zero deflection at more than four successive support points, it is evident that the same expressions vanish everywhere on any portion of the elastic curve in this region. Hence, we get

Theorem 2. If a uniform continuous beam is simply supported at the support points x = a, a+1, . . . a+n, with $n \ge 3$, and has zero deflection at all these support points, then the deflection y(x) satisfies the finite difference equation

(2.5)
$$y(x+2) + \mu y(x+1) + y(x) = 0$$

in the closed interval a≤x≤a+n.

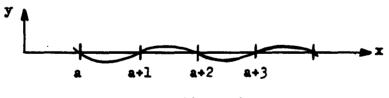


Figure 3

Since y(x) is of class C^n in this interval, equation (2.5) may be differentiated twice to yield the results:

Theorem 2a. The functions y^1 , y^n and hence the slopes, curvatures and moments satisfy also the same difference equation.

SOLUTION OF THE DIFFERENCE EQUATION

It is intuitively evident that the elastic curve of n (≥ 3) segments through n + 1 support points where the deflection is zero is uniquely determined by prescribing slopes at both ends; hence, there is a two-parameter family of such elastic curves. Let us investigate how this appears mathematically. The difference equation

$$y(x+2) + hy(x+1) + y(x) = 0$$

may be solved by letting

$$y(x) = \beta^{X}$$

whence

$$y(x+1) = \beta^{x+1} = \beta^x \beta = \beta y(x)$$

$$y(x+2) = \beta^{x+2} = \beta^x \beta^2 = \beta^2 y(x)$$

The difference equation then becomes

(3.1)
$$\beta^2 + 4\beta + 1 = 0$$

which has the two roots

$$\beta = -2 + \sqrt{3} = -1/(2 + \sqrt{3}) = -0.26795$$

$$1/\beta = -2\sqrt{3} = -1/(2 - \sqrt{3}) = -3.7320$$

The solution of a linear homogeneous difference equation is not unique. For it is easy to check that if P(x) is any periodic function of period unity.

$$P(x+1) = P(x)$$

and if y = f(x) is a solution of the difference equation

$$y(x+2) + y(x+1) + y(x) = 0$$

then

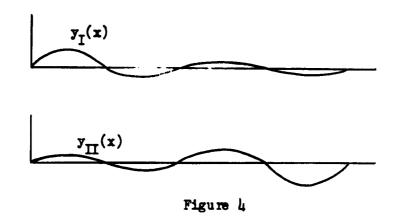
y = P(x). f(x) is also a solution. (This is the analog to the theorem in linear homogeneous differential equations that any constant times a solution is a solution).

When this result is applied to the continuous beam problem in any interval, $a \le x \le a + n$, we conclude that the solutions $y(x) = P(x)\beta^{x}$ or $P(x)/\beta^{x}$ must be cubics in any segment between successive intergers.

We therefore get the result:

Theorem 3. In any interval of n segments $(n \ge 3)$ of an elastic curve where the deflection vanishes at every support point, the deflection is a linear combination of an increasing "wave", $y(x+1) = \beta y(x)$, and a

decreasing "wave" with $y(x+1) = (1/\beta)y(x)$.

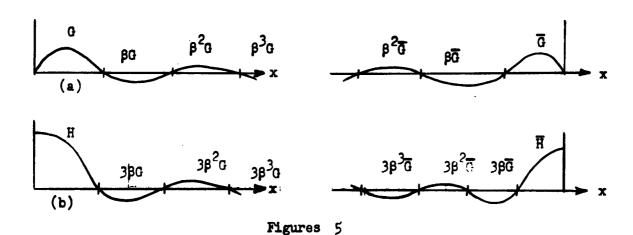


General solution is $C_T y_T(x) + C_{TT} y_{TT}(x)$.

Since $|\beta| = 1/4$ approximately, the decreasing "wave" decays to about 0.1% of its initial amplitude in five intervals, and hence can be considered to vanish for most engineering applications in five or less intervals.

THE TWO FUNDAMENTAL SOLUTIONS

It is convenient to make use of two solutions for the semi-infinite beam with zero deflections at the support points $x = 1,2,3,\ldots$ and decaying toward infinity. The first fundamental solution has zero deflection and unit slope at x = 0 (Figure 5a), while the second has unit deflection and zero slope at the origin (Figure 5b). It will be found that the solutions of particular problems can be built up from these fundamental solutions and their reflections, (Figures 5c and 5d.).



10

The first fundamental solution satisfies the difference equation over the entire range $0 \le x \le 0$ and hence is completely determined after we find the function G for the first segment $0 \le x \le 1$. The succeeding segments are the curves β G, β G, etc. Thus, the deflection, slope, and curvature at the right end of G are β times their initial values to give continuity of these functions at x = 1.

Hence the cubic for Gg

$$y = Ax^3 + Bx^2 + Cx + D$$

satisfies the initial conditions

$$y(0) = D = 0$$

$$y^{\dagger}(0) = C = 1$$

and the continuity conditions at x = 1 (after inserting values of C,D)

$$A + B + 1 = 0$$

$$3A + 2B + 1 = \beta$$

$$6A + 2B = 2B\beta$$

These three equations are linearly dependent; the solution of any pair gives

(4.1)
$$G = (\sqrt{3} - 1) x^3 - \sqrt{3} x^2 + x$$

The second fundamental solution (Figure 5b) satisfies the difference equation to the right of x = 1, so that the elastic curve to the right of x = 1 is known to be - g G, where - g is the slope at x = 1. Thus for the function H.

$$y = Ax^3 + Bx^2 + Cx + D$$

we have from initial conditions

$$y(0) = D = 1$$

$$y^1(0) = C = 0$$

From continuity with -g G at x = 1 (inserting values of C,D)

$$y(1) = A+B+1 = -gG(0) = 0$$

$$y'(1) = 3A+2B = -gG'(0) = -g$$

$$y^{*}(1) = 6A + 2B = gG^{*}(0) = 2g\sqrt{3}$$

By solving these we find that

(4.2)
$$h = (3\sqrt{3} - 4)x^3 - (3\sqrt{3} - 3)x^2 + 1$$

and that

(4.3)
$$g = 6 - 3\sqrt{3} = -3\beta$$

To find the reflected functions \overline{G} and \overline{H} referred to axes at their left ends for convenience, it is only necessary to replace x by (1-x) in 4.1 and 4.2.

We get

(4.4)
$$\overline{G}(x) = G(1-x) = (-\sqrt{3} + 1)x^3 + (2\sqrt{3} - 3)x^2 + (2 - \sqrt{3})x$$

(4.5)
$$H(x) = H(1-x) = -(3\sqrt{3} - 4)x^3 + (6\sqrt{3} - 9)x^2 + (6 - 3\sqrt{3})$$

Putting the coefficients in decimal form, we have the equations

$$G(x) = 0.73205x^3 - 1.73205x^2 + x$$

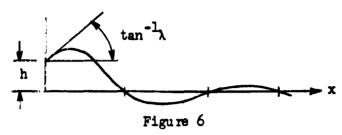
$$H(x) = 1.19615x^3 - 2.19615x^2 + 1$$

$$g = 0.80385 = -H'(1)$$

Larger graphs of the functions G and H are given at the end of this report.

APPLICATION TO SEMI-INFINITE BEAM

A. End of semi-infinite beam given a deflection h and a slope λ



The function

$$y = hH(x) + \lambda G(x)$$

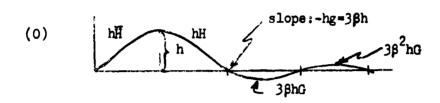
is easily seen to be the solution here, where H, G are extended beyond x = 1, as in Figure 5a and b. The slope at x = 1 is $hH'(1)+\lambda G'(1) = 3\beta h + \beta \lambda$, hence, the second segment of Fig. 6 is $(3h+\lambda)\beta G$, and the nth segment is $(3h+\lambda)\beta^{n-1}G$.

When any other jack than the end jack is moved in the semi-infinite plate, the plate shape is the sum of a zeroth order approximation, curve (0), and a first order correction, curve (1). Curve (0) is the elastic curve which would be obtained in an infinite plate extending over the range $-00 \le x < 0$; curve (1) is the reflection of this curve at the fixed end x = 0. In the case of a finite length plate, treated in the next section, the zeroth approximation, curve (0), is again the elastic curve for an infinite plate; to this must be added an infinite number of reflections of curve (0), reflected from both ends of the actual finite plate.

SEMI-INFINITE BEAMS

B. End of beam fixed at zero slope, support point at x = 1 raised by h units.





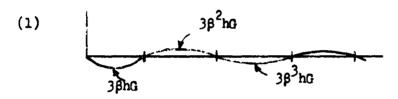


Figure 7

Curve (0) is composed of the fundamental solution H multiplied by the ordinate h starting at x = 1 and decaying to the right; the reflection function h H extends to the left of x = 1 to the origin.

Curve (1) is $-\lambda G$ starting from the origin with a slope opposite that of curve (1), where $\lambda = hg$.

The sum of curves (0) and (1) is the complete solution.

Referred to axes at the left end of each segment, the equations of the successive segments in the solution are

$$y_{1} = h \left[\overline{H}(x_{1}) + 3\beta G(x_{1}) \right]$$

$$y_{2} = h \left[H(x_{2}) + 3\beta^{2} G(x_{2}) \right]$$

$$y_{3} = h \left[3(1+\beta^{2})\beta G(x_{3}) \right]$$

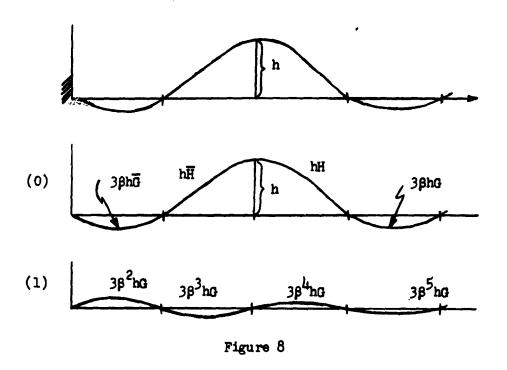
$$y_{1} = \beta y_{3},$$

$$y_{n} = \beta^{n-3}y_{3} = h 3\beta^{n-2}(1+\beta^{2})G, \text{ for } n \ge 3.$$

Clearly curve (1) can be considered to be the left wave has of curve (0) "reflected" at x = 0.

SEMI-INFINITE BEAMS

C. Left end of beam fixed, support at x = 2 raised by height h.



The solution is the sum of curves (0) and (1).

Again, curve (1) can be considered as the reflection of the "wave" to the left of the origin in figure (8).

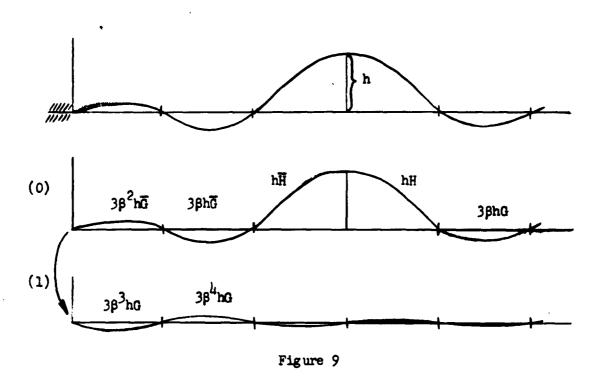
Successive segments referred to their own axis systems

$$y_1 = 3\beta h(\overline{0}+\beta G)$$

 $y_2 = h(H+3\beta^3 G)$
 $y_3 = h(H+3\beta^{l_1}G)$
 $y_{l_1} = 3h\beta(1+\beta^{l_1})G$
 $y_n = \beta^{n-l_1}y_{l_1}, n = 5, 6, ...$

SEMI-INFINITE BEAMS

D. Support at x = 3 raised.



Successive segments

$$y_1 = 3h(\beta^2 \overline{G} + \beta^3 G)$$

 $y_2 = 3h(\beta \overline{G} + \beta^4 G)$
 $y_3 = h(\overline{H} + \beta^5 G)$
 $y_4 = h(H + \beta^6 G)$
 $y_5 = 3\beta h(1 + \beta^6)G$,
 $y_n = \beta^{n-5}y_5$, $n = 6, 7, ...$

FINITE BEAM PROBLEMS WITH FIXED ENDS

a. Two spans

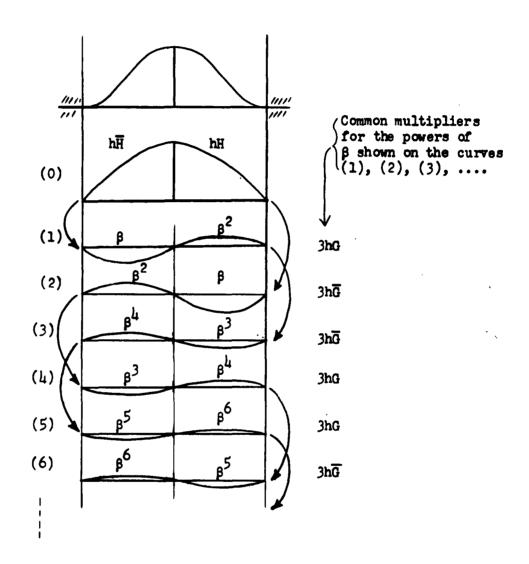


Figure 10

Curve 1 is the first reflection from the left of curve 0; the sum of these satisfies the zero slope condition on the left, but not on the right. When the reflections 2 and 3 from the right of curves 0 and 1 are added on, the zero slope condition is satisfied on the right, but not on the left. When the infinite series of reflections back and forth are added, the exact solutions are obtained.

$$y_1 = h\overline{H}(x_1) + 3\beta h \left[(1+\beta^2+\beta^{1/4}+\beta^6+ \dots)G(x_1) + (\beta+\beta^3+\beta^5+ \dots)\overline{G}(x_1) \right]$$

= $h\overline{H}(x_1) + 3\beta h (G+\beta\overline{G})/(1-\beta^2)$

This illustrates a property which will be found to be true for fixed end beams with any number of spans:

Theorem: The four terms in the reflections 1, 2, 3, and 4 for any span will always be the first terms in four geometric series of ratio β^{2n} where n is the number of spans. Since

$$A + Ar + Ar^2 + Ar^3 + \dots = A/(1-r)$$
, for $|r| \le 1$

it is only necessary to add the zero th order curve plus $1/(1-\beta^{2n})$ times the sum of the first four reflection curves 1, 2, 3 4 in order to get the exact solution.

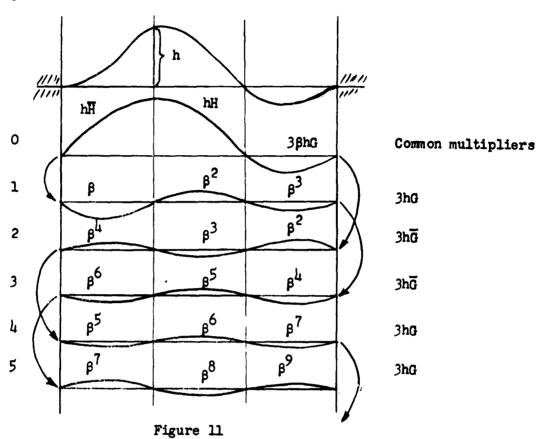
The second segment of the beam is found by adding its zeroth order term $+ V(1-\beta^{l_1})$ times the terms in the reflections 1, 2, 3, and l. In terms of its own coordinate system.

$$y_2 = hH(x_2) + 3\beta h \left[(\beta + \beta^3)G(x_3) + (1+\beta^2)\overline{G}(x_2) \right] / (1-\beta^{\frac{1}{4}})$$

or

$$y_2 = hH(x_2) + 3\beta h \left[\beta G(x_2) + \overline{G}(x_2)\right] / (1-\beta^2).$$

b. Three-Span beam



In this diagram, common multipliers for the terms in the first four reflection are placed at the right while only the proper powers of β are placed on the corresponding segments.

The solution is

$$y_{1} = h\overline{H}(x_{1}) + 3\beta h \left[(1+\beta^{4})G(x_{1}) + (\beta^{3}+\beta^{5})\overline{G}(x_{1}) \right] / (1-\beta^{6})$$

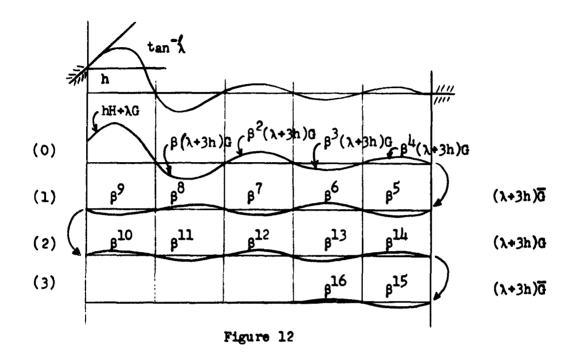
$$y_{2} = hH(x_{2}) + 3\beta h \left[(\beta+\beta^{5})G(x_{2}) + (\beta^{2}+\beta^{4})\overline{G}(x_{2}) \right] / (1-\beta^{6})$$

$$y_{3} = 3\beta hG(x_{3}) + 3\beta h \left[(\beta^{2} + \beta^{6}) G(x_{3}) + (\beta + \beta^{3}) \overline{G}(x_{3}) \right] / (1-\beta^{6})$$

$$= 3\beta h \left[(1+\beta^{2}) G(x_{3}) + (\beta + -\beta^{3})\overline{G}(x_{3}) \right] / (1-\beta^{6})$$

$$= 3\beta h (1+\beta^{2}) \left[G(x_{3}) + \beta G(x_{2}) \right] / (1-\beta^{6})$$

c. Five-span Beam with Prescribed Initial Height and Slope



The zero th order curve shown satisfies left end boundary conditions, hence only two reflections, curves 1 and 2, need be considered to get the first terms of the infinite series (geometrical) and hence to obtain the exact solution.

For the 5 span beam shown:

$$y_{1} = hH + \lambda G + \beta(\lambda + 3h) \left[\beta^{9}G + \beta^{8}\overline{G} \right]$$

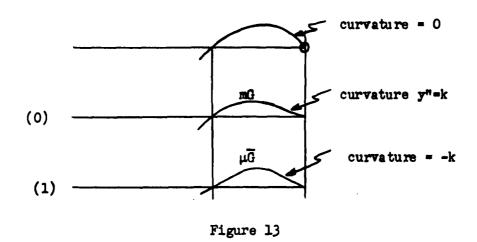
$$y_{2} = \beta(\lambda + 3h)G + \beta(\lambda + 3h) \left[\beta^{10}G + \beta^{7}\overline{G} \right] / (1 - \beta^{10}) = +\beta(\lambda + 3h) \left[G + \beta^{7}\overline{G} \right] / (1 - \beta^{10})$$

$$y_{3} = \beta(\lambda + 3h) \left[\beta G + \beta^{6}\overline{G} \right] / (1 - \beta^{10})$$

$$y_{4} = \beta(\lambda + 3h) \left[\beta^{2}G + \beta^{5}\overline{G} \right] / (1 - \beta^{10})$$

$$y_{5} = \beta(\lambda + 3h) \left[\beta^{2}G + \beta^{5}\overline{G} \right] / (1 - \beta^{10})$$

REFLECTION FROM SIMPLY-SUPPORTED END



Suppose the zeroth order approximation ends with the terms mG, and the reflection curve 1 ends with the terms μG ; then in order that the sum of curves 0 and 1 satisfy the boundary condition y^{*} ! = 0, we must have

$$mG^{11}(1)+\mu\overline{G}^{11}(1)=0$$

 $m(6\sqrt{3}-6-2\sqrt{3})+\mu(-6\sqrt{3}+6+4\sqrt{3}-6)=0$

Therefore $\mu = + k(end)/2 \sqrt{3}$

=
$$(4\sqrt{3} - 6)m/2\sqrt{3} = (2 - \sqrt{3})m = -\beta m$$

Theorem: The reflection on the right hand end starts with k $\overline{G}/2\sqrt{3}$ where k is the corvature at the right on the zero order approximation. If the right of the zero order curve is mG, then the right end of the reflection curve is $(-m\beta G)$. The reflection from a simply supported end is with no change in sign of y, since $-\beta$ is positive. The same theorem holds at a simply supported left end, with the words left and right and the symbols G and G interchanged.

As a consequence, if the zero th order approximation ends in a multiple of \overline{G} (at the left) or a multiple of G (on the right), the exact solution has twice the end slope of the zero approximation.

The same principle of multiplying by $-\beta$ instead of by β holds, of course, for all the infinite series of reflections needed in a finite beam with pin-supported (simply supported) ends.

See Fig. 14 for an example of a 3-span beam with the left end pin-connected and the right end fixed.

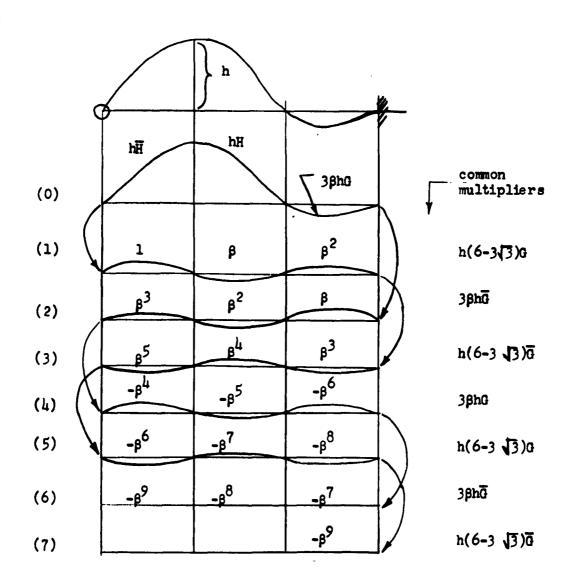


Figure 14

$$y_{1} = h\overline{H} + (6-3\sqrt{3}) h \left[(1-\beta^{6} + \beta^{12} - ...)G + \beta^{5} - \beta^{11} + \beta^{17} - ...) \right]$$

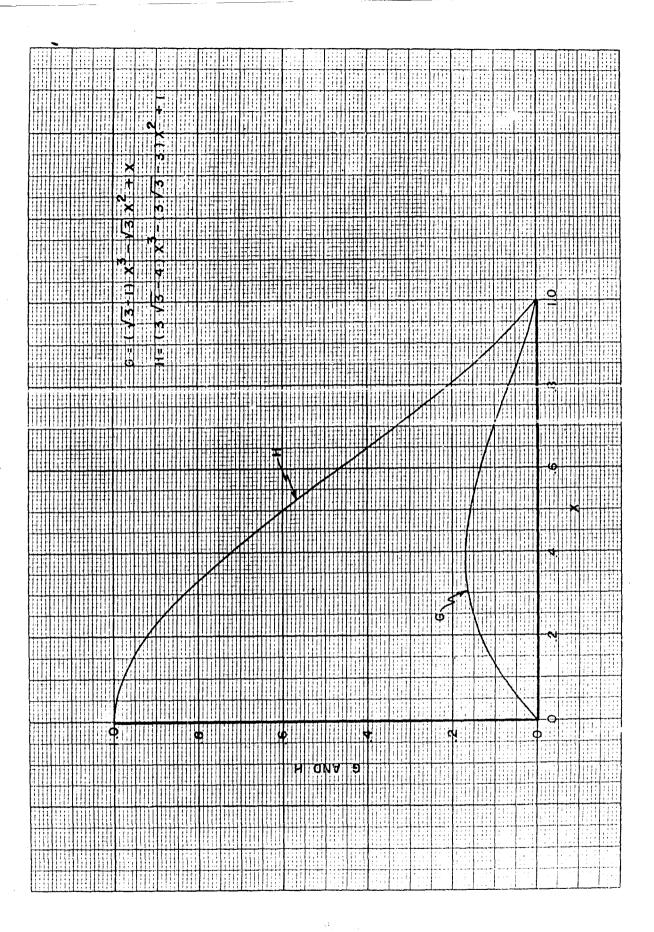
$$+3\beta h \left[(\beta^{3} - \beta^{9} + \beta^{15} - ...) + (\beta^{4} - \beta^{10} + \beta^{16} - ...) \right]$$

$$= h\overline{H} + (6-3\sqrt{3}) h (G + \beta^{5}\overline{G}) / (1+\beta^{6}) + 3\beta h (\beta^{3}\overline{G} + \beta^{4}G) / (1+\beta^{6})$$

Turner L. Smith

Again the first four reflection curves are enough, but the geometric series are alternating with the ratio $-\beta^0$; hence their first terms must be divided by $1+\beta^0$ to obtain their sums.

TURNER L. SMITH



DISTRIBUTION LIST

6 Chief of Ordnance Department of the Army Washington 25, D. C. Attn: ORDTB - Bal Sec Documents Service Center Knott Building Dayton 2, Ohio Attn: DSC - SA L. Canadian Joint Staff - ORDTB for distribution CRDTB for distribution	
lo British - ORDTB for distribution La Canadian Joint Staff - la Commanding Officer and Direct ORDTB for distribution La Chief, Bureau of Ordnance Department of the Navy Washington 25, D. C. Attn: Aeromechanics Division Department of the Navy Washington 25, D. C. La Chief, Bureau of Ordnance Attn: Re3 La Chief, Bureau of Ordnance Attn: Aeromechanics Division Department of the Navy Washington 25, D. C. La California Institute of Technology Pasadena, California Attn: Mr. Frank Goddard	
ORDTB for distribution David Taylor Model Basin Washington 7, D. C. Attn: Aeromechanics Division Department of the Navy Washington 25, D. C. Attn: Re3 ASTIA Reference Center David Taylor Model Basin Washington 7, D. C. Attn: Aeromechanics Division California Institute of Tec Jet Propulsion Laboratory Pasadena, California Attn: Mr. Frank Goddard	
4 Chief, Bureau of Ordnance Attn: Aeromechanics Division Department of the Navy Washington 25, D. C. 1 California Institute of Technology Pasadena, California Attn: Mr. Frank Goddard	ector
Attn: Re3 Jet Propulsion Laboratory Pasadena, California ASTIA Reference Center Attn: Mr. Frank Goddard	
4 ASTIA Reference Center Attn: Mr. Frank Goddard	cunorogy
Washington 25, D. C. 1 New York University New York, New York Commander Attn: Professor J. J. Stol	leam
2 Commander Attn: Professor J. J. Stole Naval Proving Ground Dahlgren, Virginia 1 Carnegie Institute of Techn	
Pittsburgh, Pennsylvania Commander Attn: Professor Saibel Naval Ordnance Laboratory	
White Oak 1 Arnold Engineering Developm Silver Spring 19, Maryland Center Tullahoma, Tennessee	ment
1 Commander Attn: Mr. R. Smelt Naval Ordnance Test Station	•
Inyokern 1 Sandberg-Serrell Corporation P. O. China Lake, California 19 North Cataline Avenue Attn: Technical Library Pasadena 1, California	n
1 Superintendent 1 Bethlehem Steel Company Naval Postgraduate School Shipbuilding Division Monterey, California Cuincy, Massachusetts Attn: Mr. A. L. Gardner	
l Director Air University Library l Mr. A. E. Puckett Maxwell Air Force Base, Alabama Hughes Aircraft Company Culver City, California	

DISTRIBUTION LIST

No. of Copies	Organization
1	Professor Howard W. Emmons Division of Applied Science Harvard University Cambridge 38, Massachusetts
1	Professor Francis H. Clauser Department of Aeronautical Engineering The Johns Hopkins University Baltimore 18, Maryland
1	Dr. Hugh L. Dryden National Advisory Committee for Aeronautics 1724 F Street, NW Washington 25, D. C.
1	Professor John Cell Department of Mathematics Box 5638 State College Station Raleigh, North Carolina
1	University of Illinois Department of Theory and Applied Mechanics Urbana, Illinois Attn: Mr. P. Schwab